

# Wireless Power Transmission

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and its applications for powering Drones

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# Summary

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# Introduction

- **Unmanned Aerial Vehicles (UAV):**
  - non-crewed aircrafts that can either be **autonomous** or **remotely controlled**;
  - associated with **several successful applications**;



Fig. 1: Parrot AR.Drone 2.0 [1].

# Introduction

**Major drawback**



Reduced autonomy



Fig. 1: Parrot AR.Drone 2.0 [1].

# Introduction

- **We propose:**

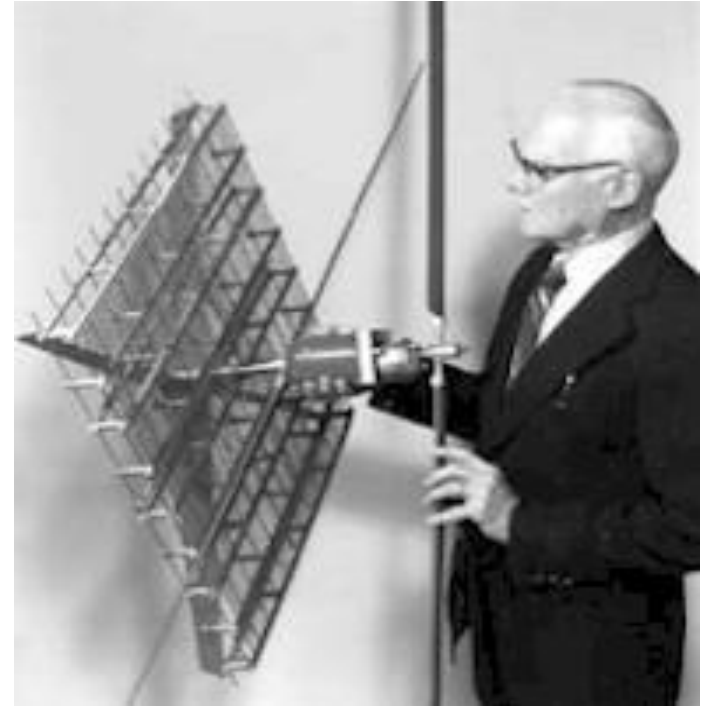


Fig. 2: Representation of a drone charging with resort to microwave power transmission.

# History Of Wireless Power Transmission



**Fig. 3: Nikola Tesla in his laboratory [2].**



**Fig. 4: William C. Brown holding the wireless power helicopter [3].**

[2] **Ahead of his time: the genius of Nikola Tesla** (February, 2013) Retrieved from: <http://lucidthoughts.com.au/wordpress/?p=1268/>

[3] **William C. Brown** (November, 2014). Retrieved from: <http://mainland.cctt.org/istf2008/brown.asp>

# Description Of The Proposed System

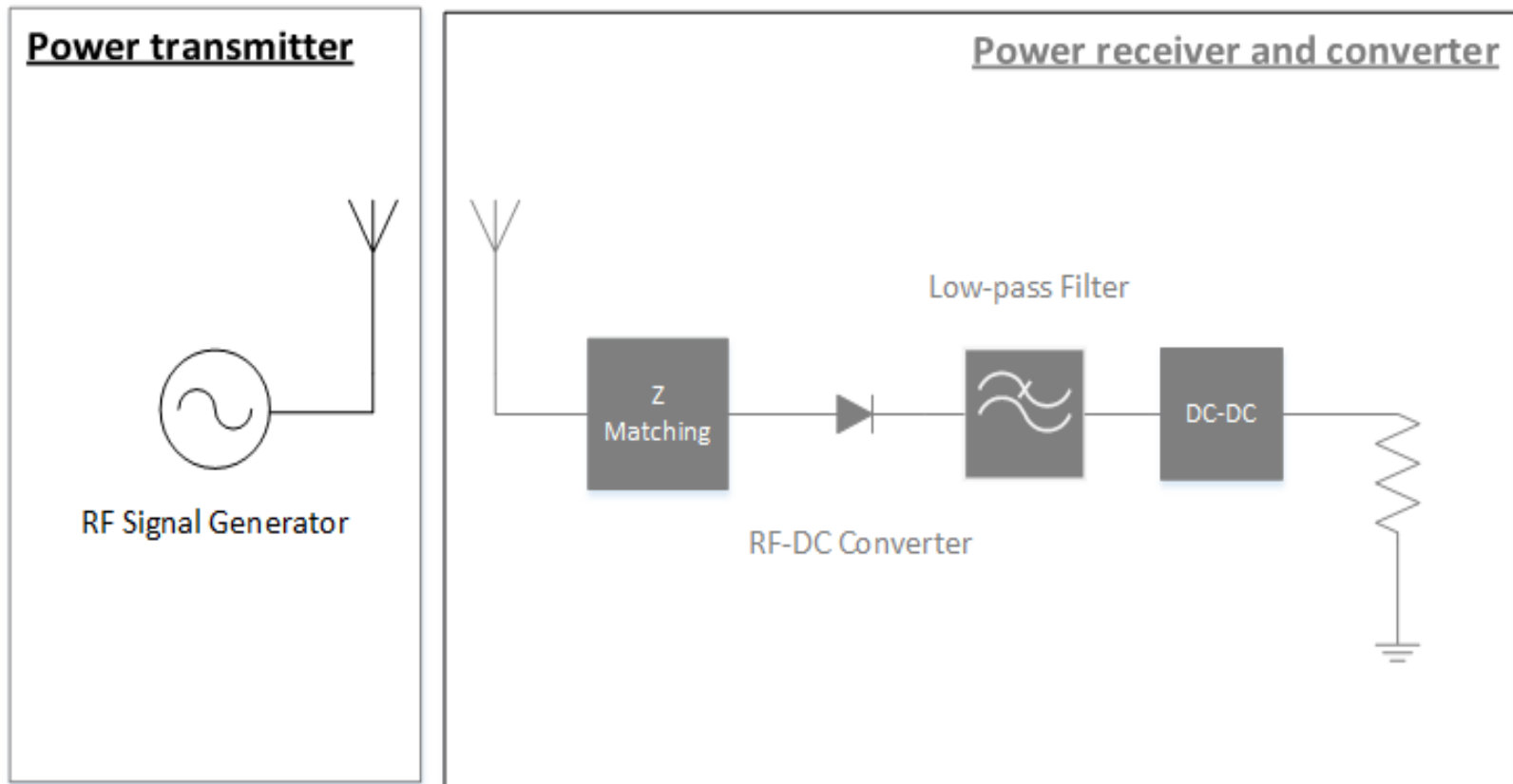


Fig. 6: Wireless power transfer system consisting of a transmitter and receiver section.

# Description Of The Proposed System

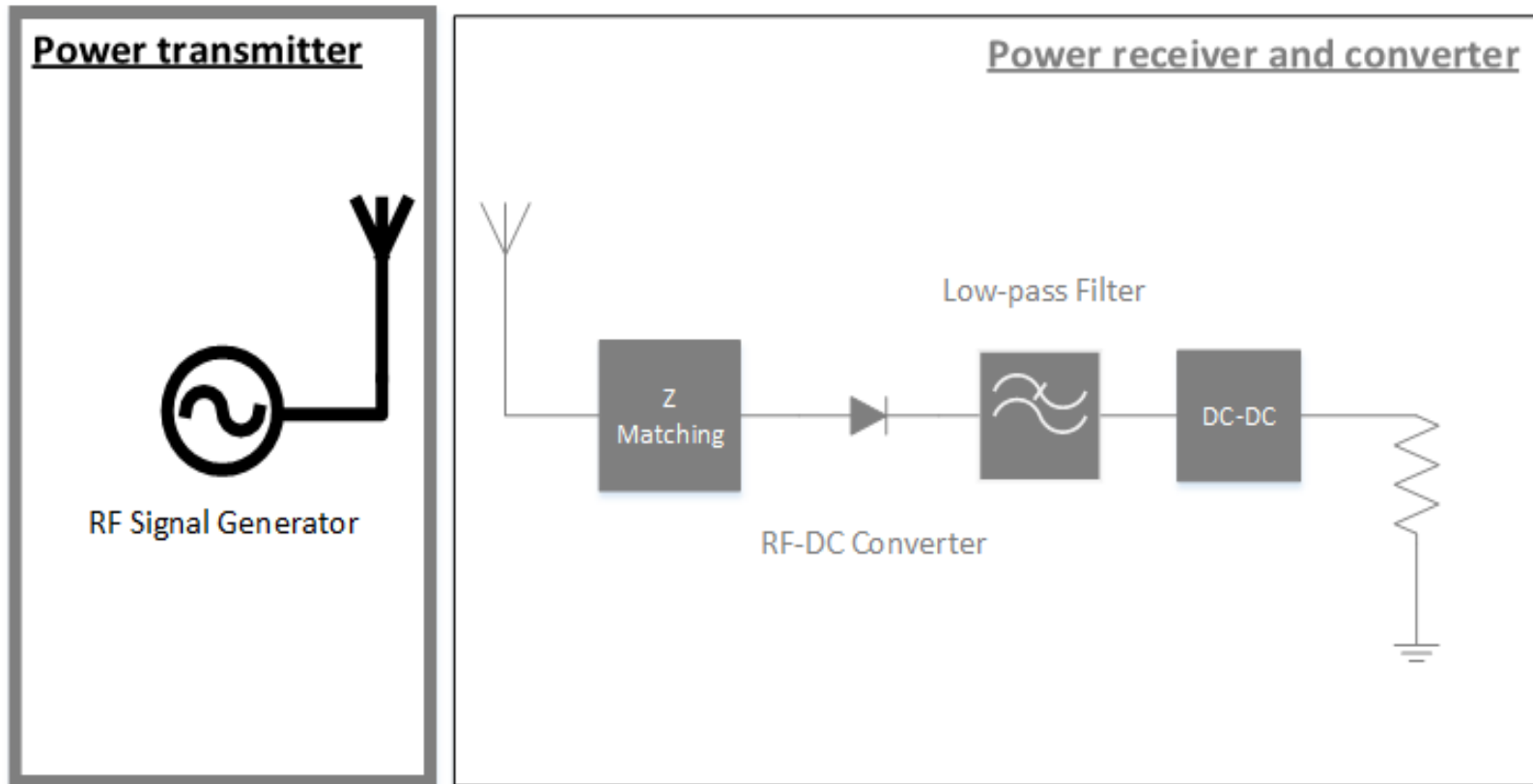


Fig. 6.1: Power transmitter.



# Transmitter

- Linearly polarized, 16 element array;
- Total dimension: 14 x 14 cm;

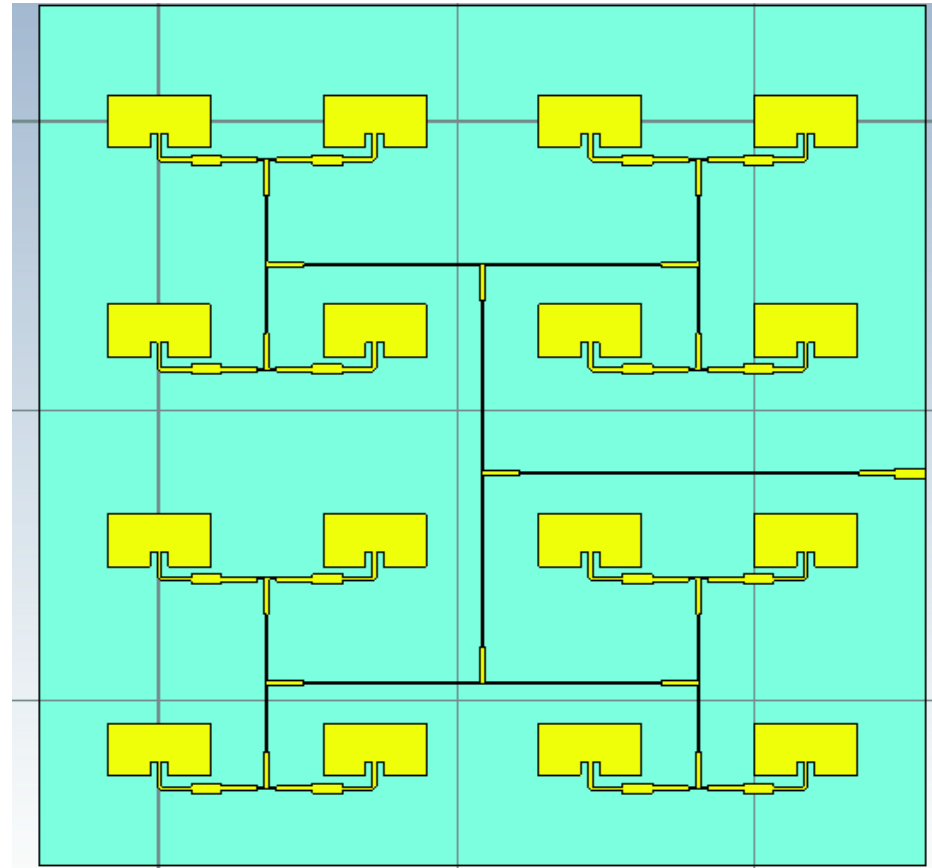
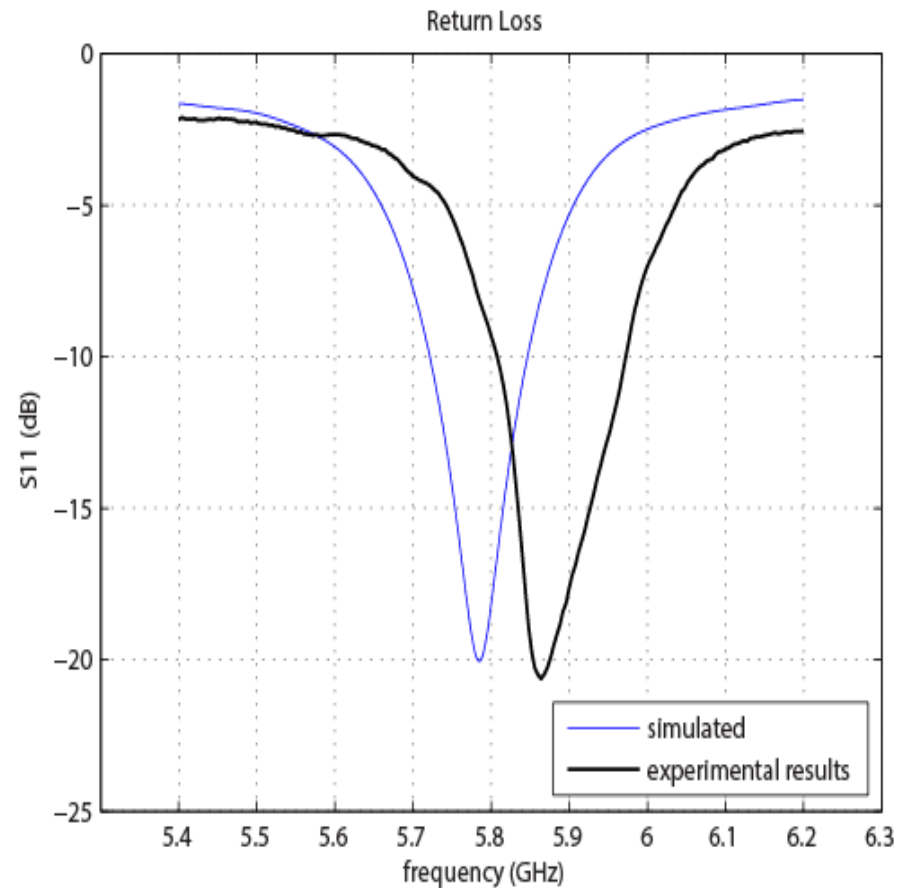


Fig. 7: Final array layout.

# Transmitter

- Measured return loss of 9.8 dB at 5,8 GHz.



**Fig. 8: Comparison between the measured and simulated S<sub>11</sub> of the full array.**

# Transmitter

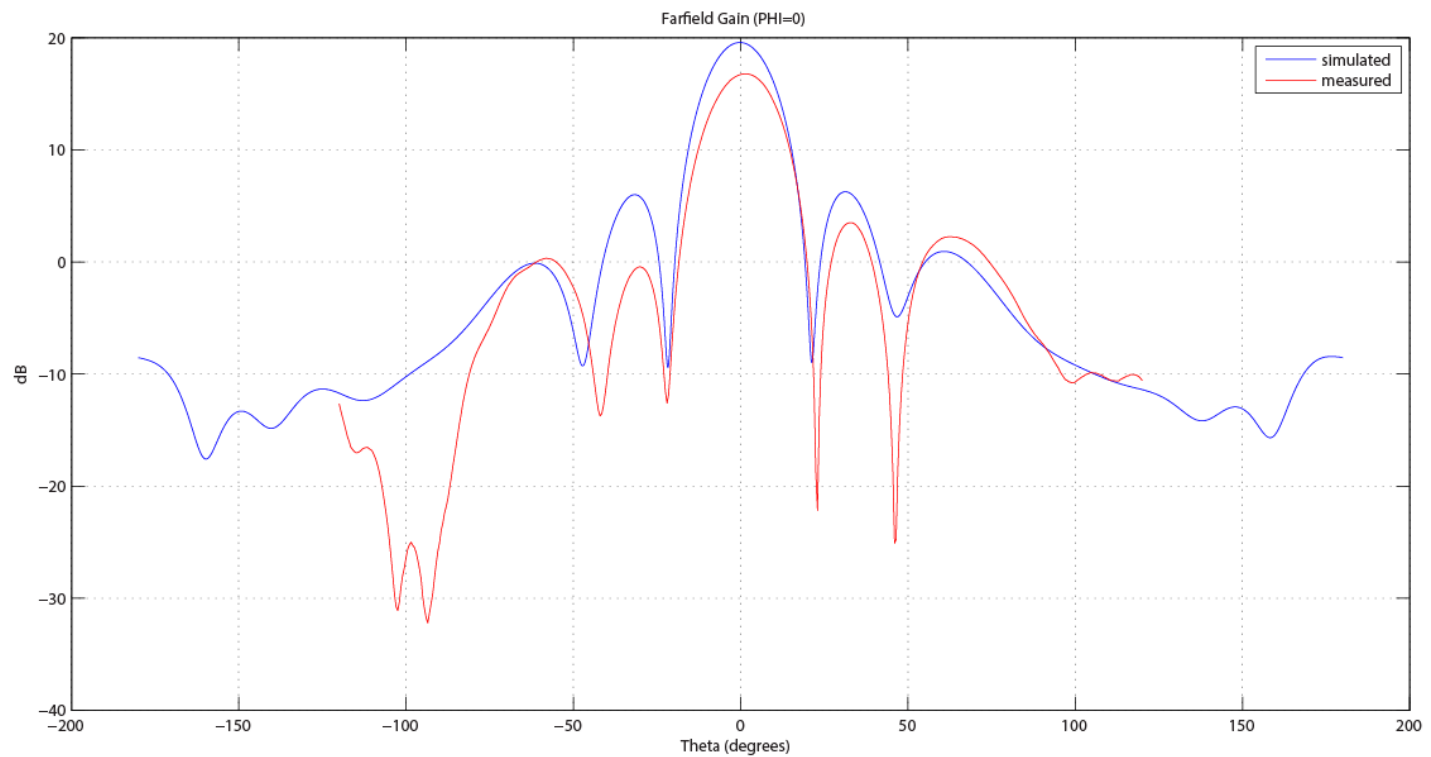


Fig. 9: Simulated and measured gain variation with theta of the 4x4 patch antenna array.

# Description Of The Proposed System

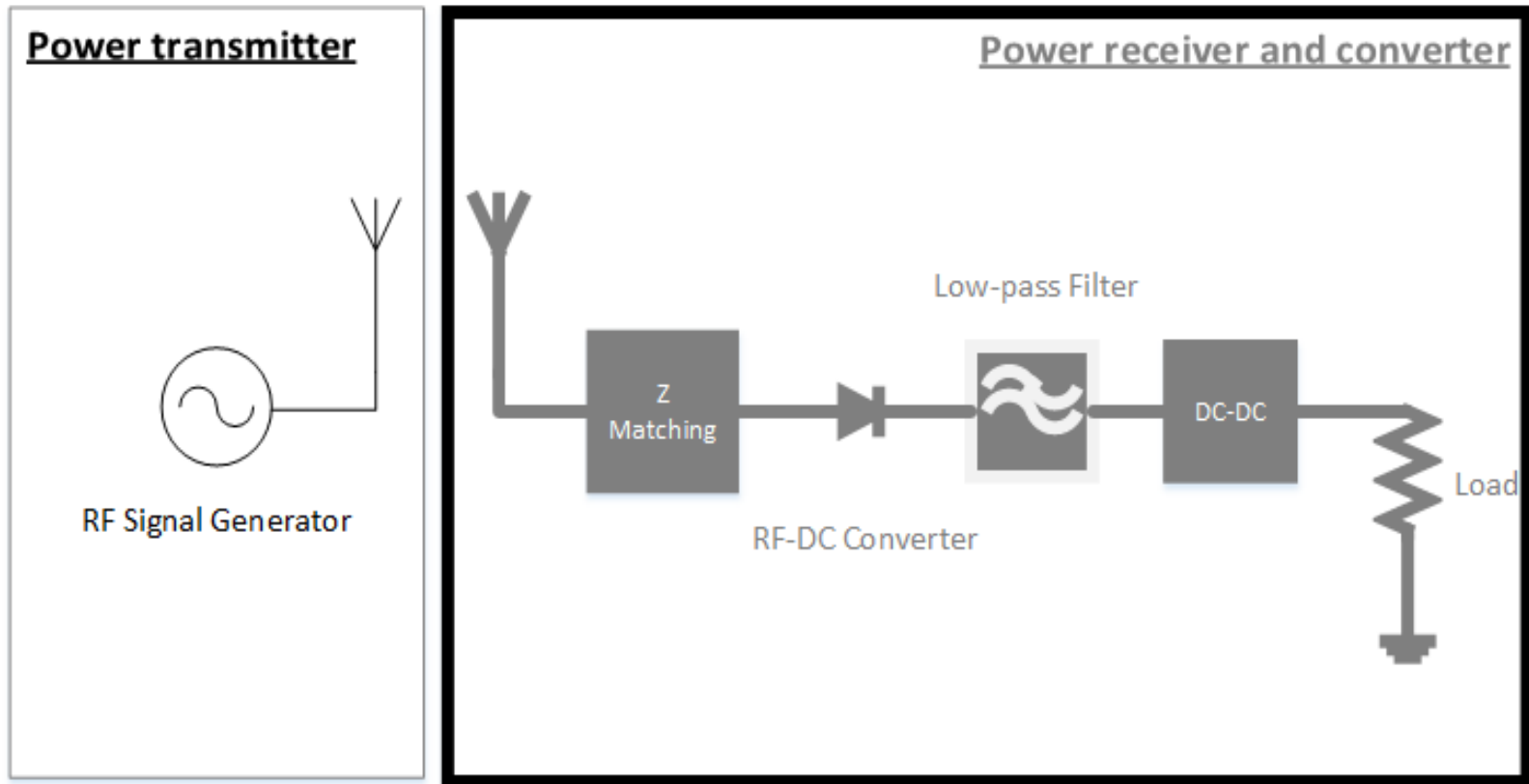


Fig. 6.2: Power receiver and converter.

# Antennas

- Patch antennas:
  - Can be easily applied to the hull of the drone;

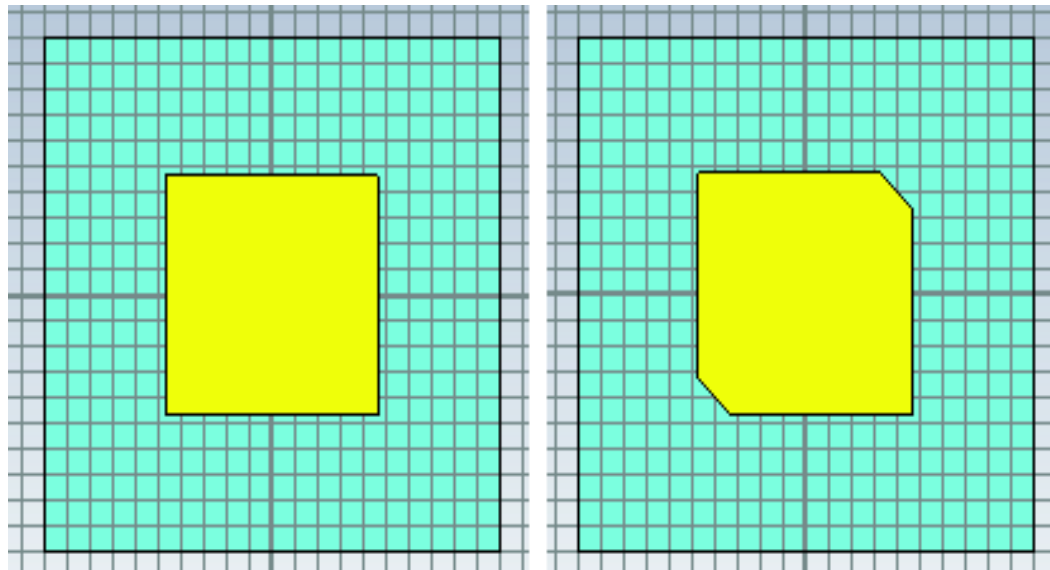
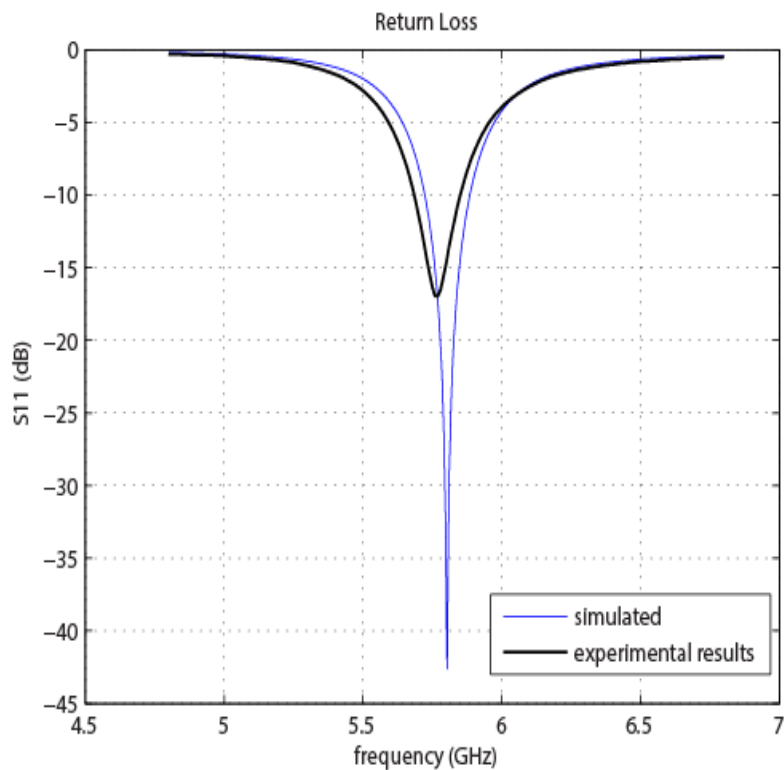
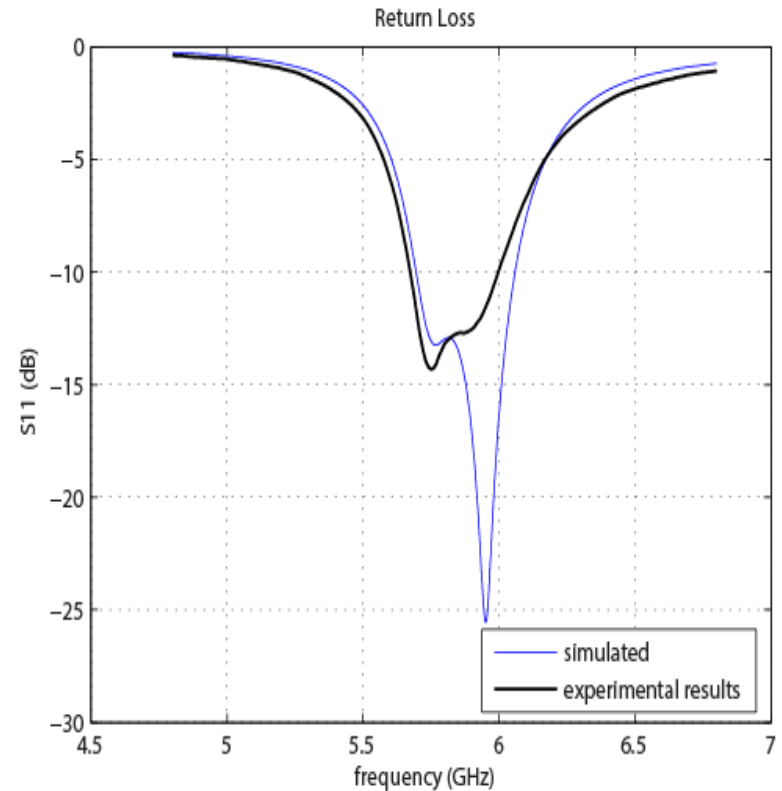


Fig. 11: Receiver antennas.

# Antennas



**Fig. 12: Comparison between simulated and measured value for the linearly polarized square patch.**



**Fig. 13: Comparison between simulated and measured values for the right hand circularly polarized patch.**

## RF-DC

- Single shunt rectifier:
  - Maximum expected conversion efficiency of 50 %;
- For the low pass filter a 100 pF capacitor was chosen;
- The HUBSAN X4 drone requires 6,6 W and was considered as a variable resistor;

# RF-DC

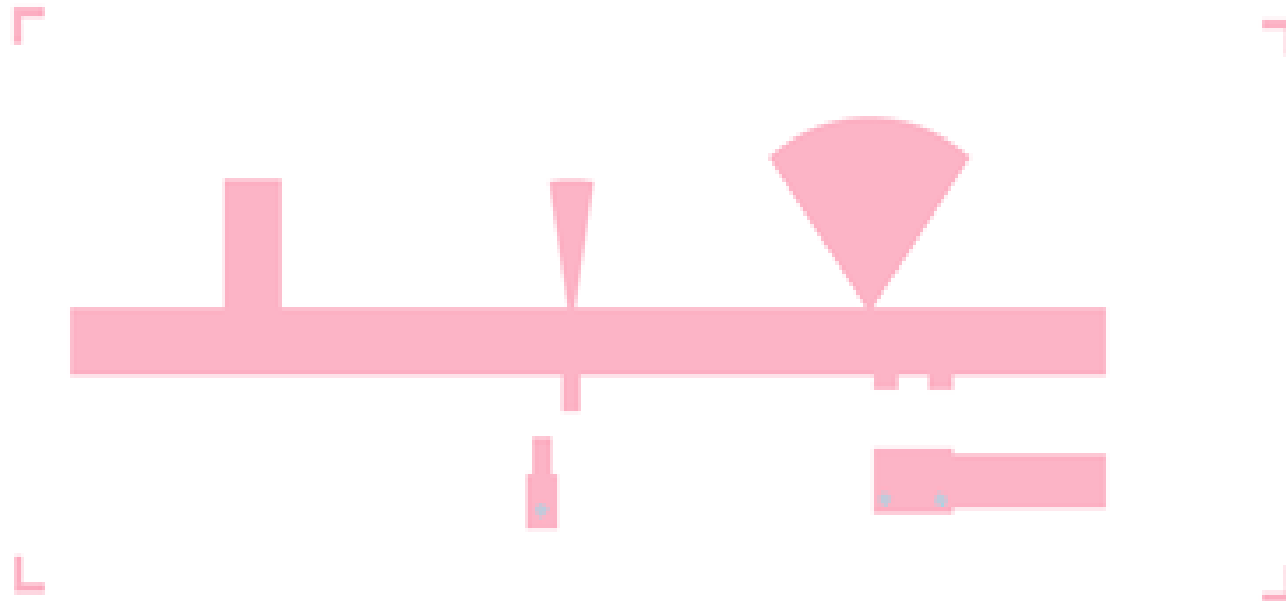
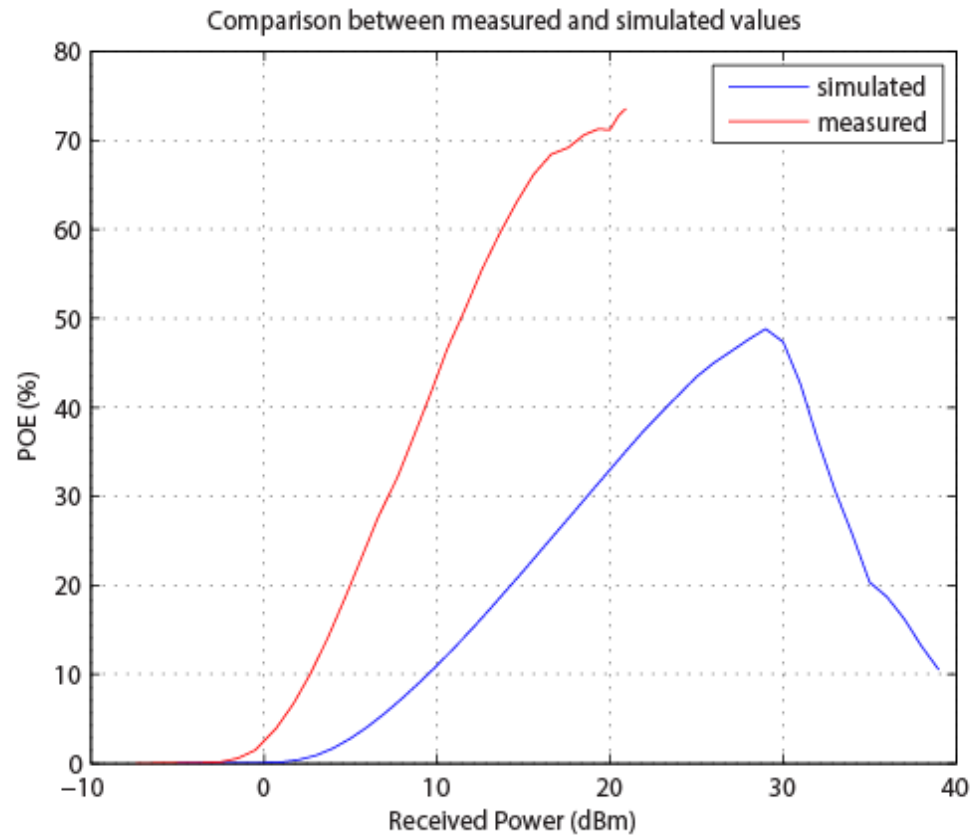


Fig. 14: Single shunt rectifier circuit layout.



# RF-DC



**Fig. 16: Simulated versus measured POE of the Momentum simulated RF-DC Converter.**



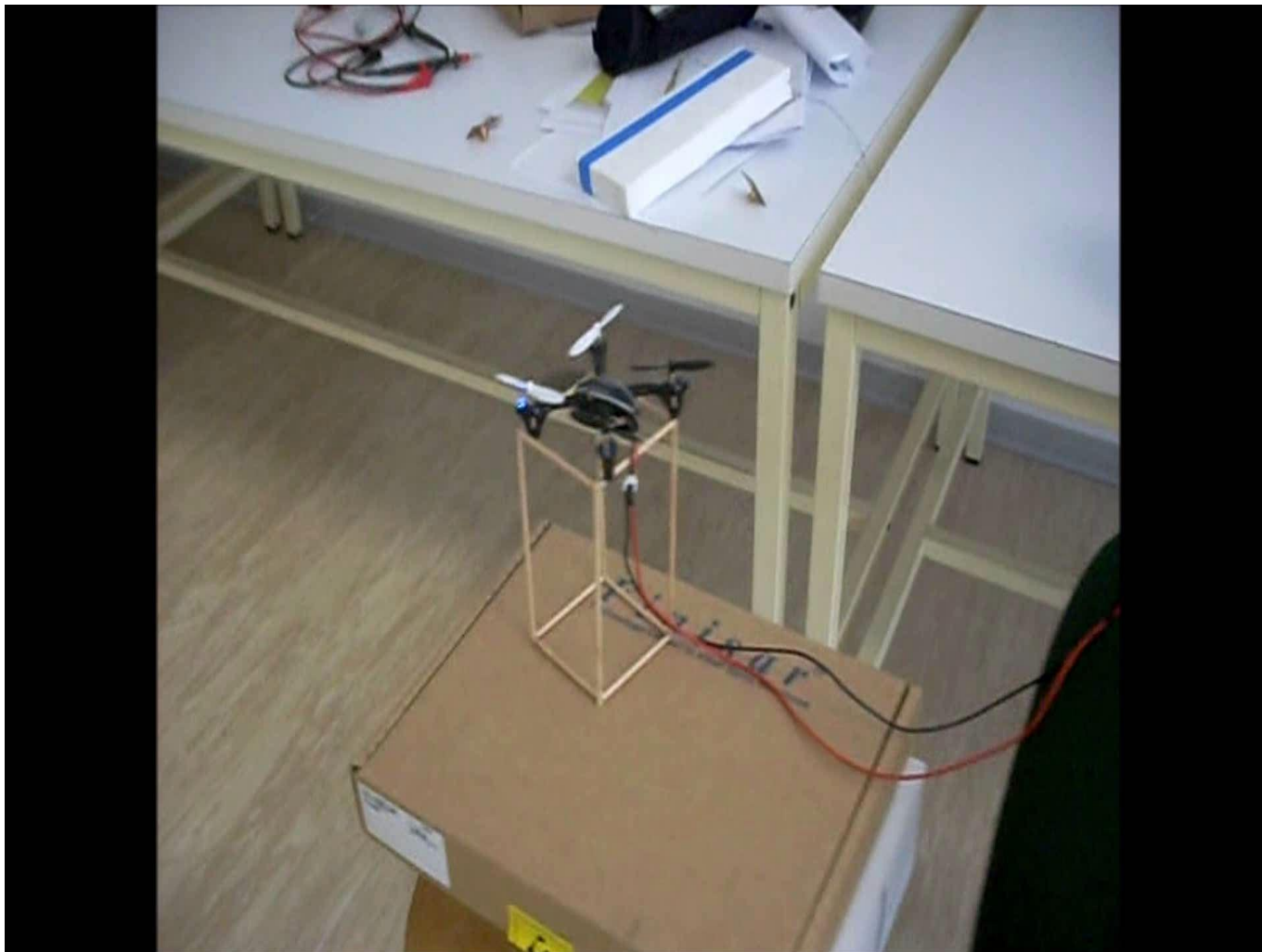
# Summary

- A 16 element array was developed for the transmitter;
- 2 differently polarized patches were printed for the receiver;
- A shunt rectifier was designed for RF-DC conversion;

# Experiments With The Drone

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# Experiments With The Drone





# Conclusion

- A wireless power system was proposed to tackle the reduced autonomy of drone;
- Several elements of this system were designed, implemented and tested;
- All the shown components present potential in being further implemented in fully or temporarily charging an unmanned aerial vehicle;

# Acknowledgements

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# Thank you for you attention!

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Are there any questions?

